

Heat Transfer
DM23815

Chapter 3. One-dimensional steady-state conduction

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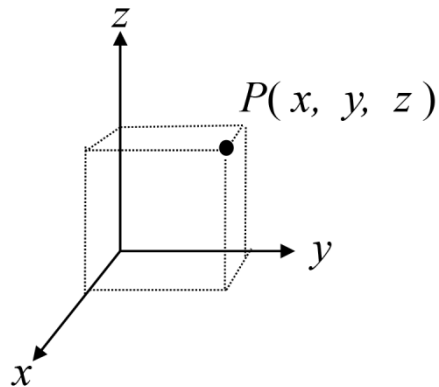


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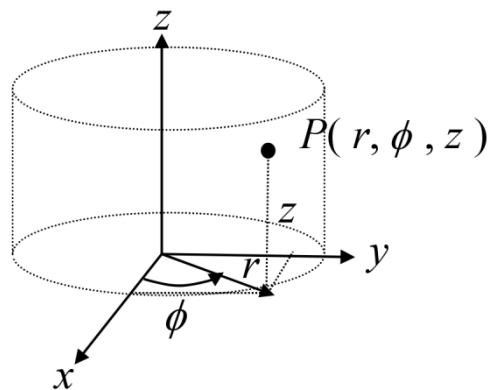
3. Introduction

Coordinate system

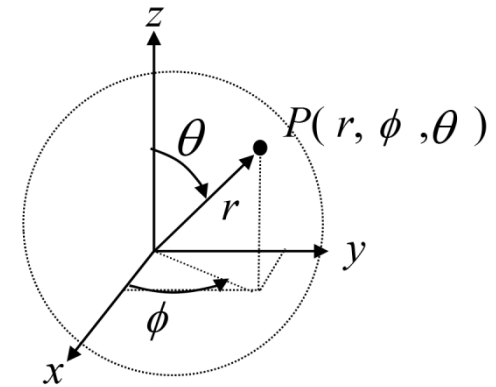
Cartesian coordinates



Cylindrical coordinates



Spherical coordinates



- One dimensional system

Temperature gradients exist along a single coordinate system, and heat transfer occurs **exclusively** in common (planar, cylindrical, and spherical) geometries.

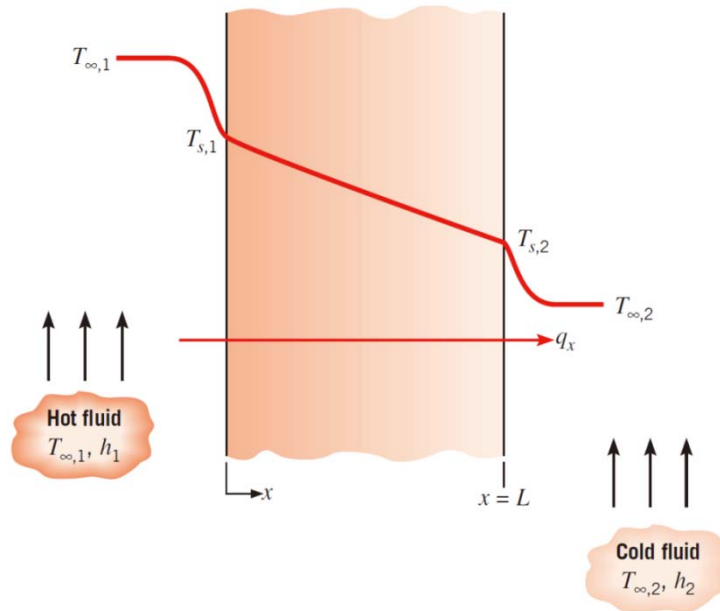
$$\frac{\partial T}{\partial x} \gg \frac{\partial T}{\partial y}, \frac{\partial T}{\partial x} \gg \frac{\partial T}{\partial z}$$

Under steady-state, one-dimensional conditions with no heat generation

$$\frac{\partial}{\partial x} \left(k \frac{\partial T}{\partial x} \right) + \frac{\partial}{\partial y} \left(k \frac{\partial T}{\partial y} \right) + \frac{\partial}{\partial z} \left(k \frac{\partial T}{\partial z} \right) + \dot{q} = \rho C \frac{\partial T}{\partial t} \quad \longrightarrow \quad \frac{d}{dx} \left(k \frac{dT}{dx} \right) = 0$$

3.1 Plain Wall

Temperature Distribution



constant thermal conductivity

$$\frac{d}{dx} \left(k \frac{dT}{dx} \right) = 0$$

$$\int d \left(\frac{dT}{dx} \right) = \int 0 dx \Rightarrow \frac{dT}{dx} = C_1$$

$$\int dT = \int C_1 dx$$

To obtain two unknown constants, two boundary conditions $T(0) = T_{s,1}$ and $T(L) = T_{s,2}$ are necessary.

$$\text{at } x = 0 \quad T_{s,1} = C_2$$

$$T(x) = (T_{s,2} - T_{s,1}) \frac{x}{L} + T_{s,1}$$

For one-dimensional, steady-state with no heat generation and constant conductivity, the **temperature** varies **linearly** with x .

$$q_x = -kA \frac{dT}{dx} = \frac{kA}{L} (T_{s,1} - T_{s,2})$$

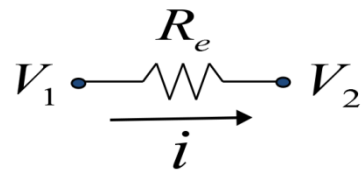
$$q_x'' = \frac{q_x}{A} = \frac{k}{L} (T_{s,1} - T_{s,2})$$

3.1 Plain Wall

▪ Analogy Analysis

Resistance: Ratio of a driving potential to the corresponding transfer rate

Electricity



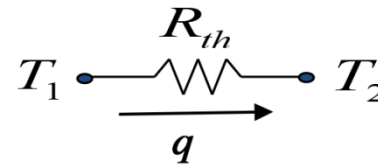
Voltage: $(V_2 - V_1)$

Current: i

$$R_e = \frac{V_1 - V_2}{i}$$

Resistance: R_e

Heat transfer



Thermal potential: $(T_2 - T_1)$

Heat flow: q

$$R_{th} = \frac{T_1 - T_2}{q}$$

Thermal resistance: R_{th}

Conduction Case

$$q_x = \frac{kA}{L}(T_{s,1} - T_{s,2})$$

$$R_{t,cond} \equiv \frac{T_{s,1} - T_{s,2}}{q_x} = \frac{L}{kA}$$

Convection Case

$$q = hA(T_s - T_\infty)$$

$$R_{t,conv} \equiv \frac{T_s - T_\infty}{q} = \frac{1}{hA}$$

Radiation Case

$$q = \varepsilon\sigma A(T_s^4 - T_{sur}^4)$$

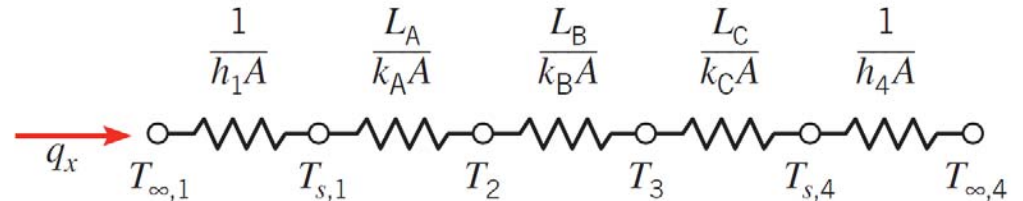
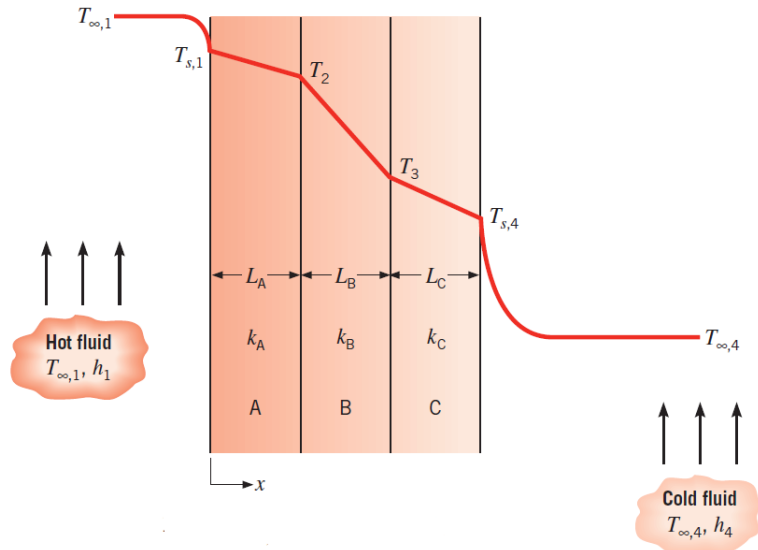
$$q = \varepsilon\sigma A(T_s^2 + T_{sur}^2)(T_s + T_{sur})(T_s - T_{sur})$$

$$q = h_{rad}A(T_s - T_{sur})$$

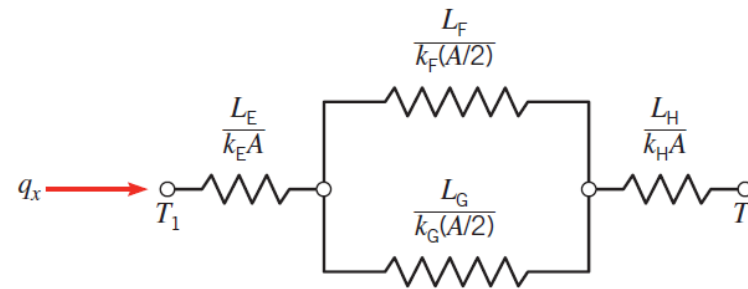
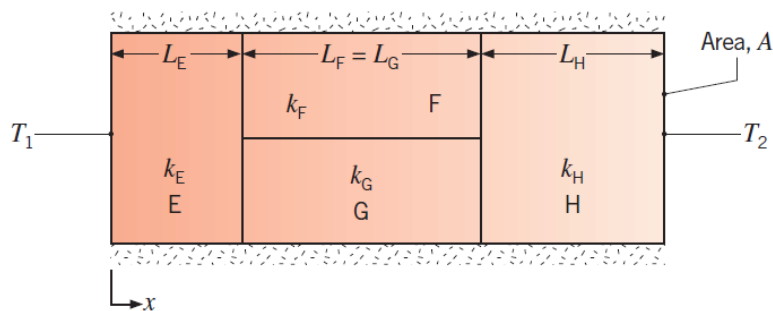
$$R_{t,rad} \equiv \frac{T_s - T_{sur}}{q} = \frac{1}{h_{rad}A}$$

3.1 Plain Wall

Steady heat conduction in multilayer plane wall



$$q_x = \frac{T_{\infty,1} - T_{\infty,4}}{[(1/h_1 A) + (L_A/k_A A) + (L_B/k_B A) + (L_C/k_C A) + (1/h_4 A)]}$$



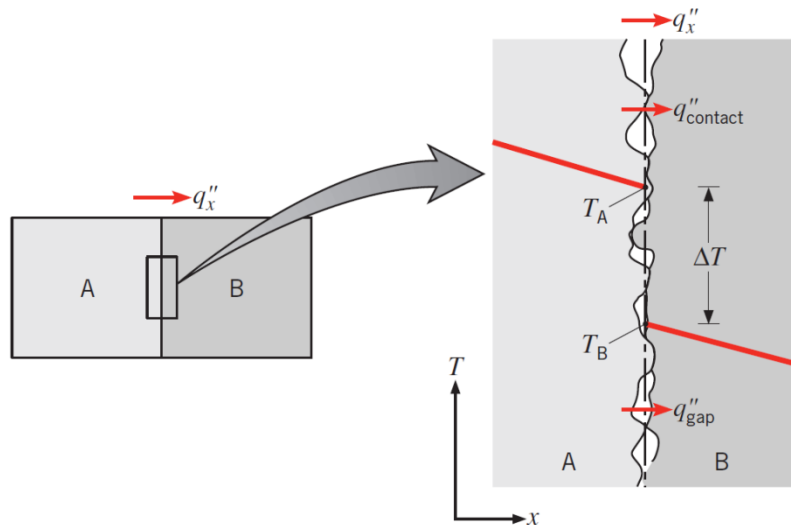
Overall heat transfer coefficient(열관류계수), U

$$q_x = UA \Delta T$$

$$U = \frac{1}{R_{\text{tot}} A} = \frac{1}{[(1/h_1) + (L_A/k_A) + (L_B/k_B) + (L_C/k_C) + (1/h_4)]}$$

3.1 Plain Wall

Thermal Contact Resistance



- The existence of a finite contact resistance is due principally to **surface roughness effects**.
- Contact spots are interspersed with gaps that are, in most instances, **air filled**. Heat transfer is therefore due to **conduction** across the **actual contact area** and to **conduction** and/or **radiation** across **the gaps**.
- The contact resistance may also be reduced by selecting an **interfacial fluid** of large thermal conductivity.

Thermal contact resistance for (a) metallic interfaces under vacuum conditions and (b) aluminum interface

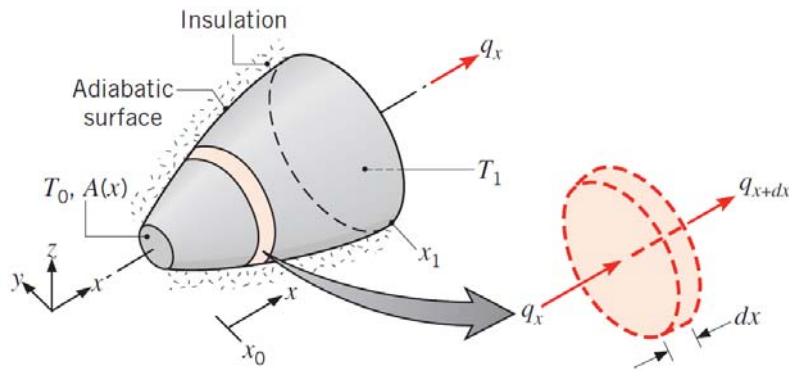
Thermal Resistance, $R''_{t,c} \times 10^4 \text{ (m}^2 \cdot \text{K/W)}$

(a) Vacuum Interface			(b) Interfacial Fluid	
Contact pressure	100 kN/m ²	10,000 kN/m ²	Air	2.75
Stainless steel	6–25	0.7–4.0	Helium	1.05
Copper	1–10	0.1–0.5	Hydrogen	0.720
Magnesium	1.5–3.5	0.2–0.4	Silicone oil	0.525
Aluminum	1.5–5.0	0.2–0.4	Glycerine	0.265

Thermal resistance of representative solid/solid interfaces

Interface	$R''_{t,c} \times 10^4 \text{ (m}^2 \cdot \text{K/W)}$
Silicon chip/lapped aluminum in air (27–500 kN/m ²)	0.3–0.6
Aluminum/aluminum with indium foil filler (~100 kN/m ²)	~0.07
Stainless/stainless with indium foil filler (~3500 kN/m ²)	~0.04
Aluminum/aluminum with metallic (Pb) coating	0.01–0.1
Aluminum/aluminum with Dow Corning 340 grease (~100 kN/m ²)	~0.07
Stainless/stainless with Dow Corning 340 grease (~3500 kN/m ²)	~0.04
Silicon chip/aluminum with 0.02-mm epoxy	0.2–0.9
Brass/brass with 15- μm tin solder	0.025–0.14

3.2 Alternative Conduction Analysis



- Under, **steady-state** conditions with **no heat generation** and no **heat loss** from the sides, heat transfer rate q_x must be a constant independent of x .
- Even if the area varies with position $A(x)$ and the thermal conductivity varies with temperature $k(T)$, $q_x = q_{x+dx}$.

$$q_x \int_{x_0}^x \frac{dx}{A(x)} = - \int_{T_0}^T k(T) dT \quad T_0 \text{ is known,} \quad T = T_1 \text{ at some } x = x_1 \text{ is known.}$$

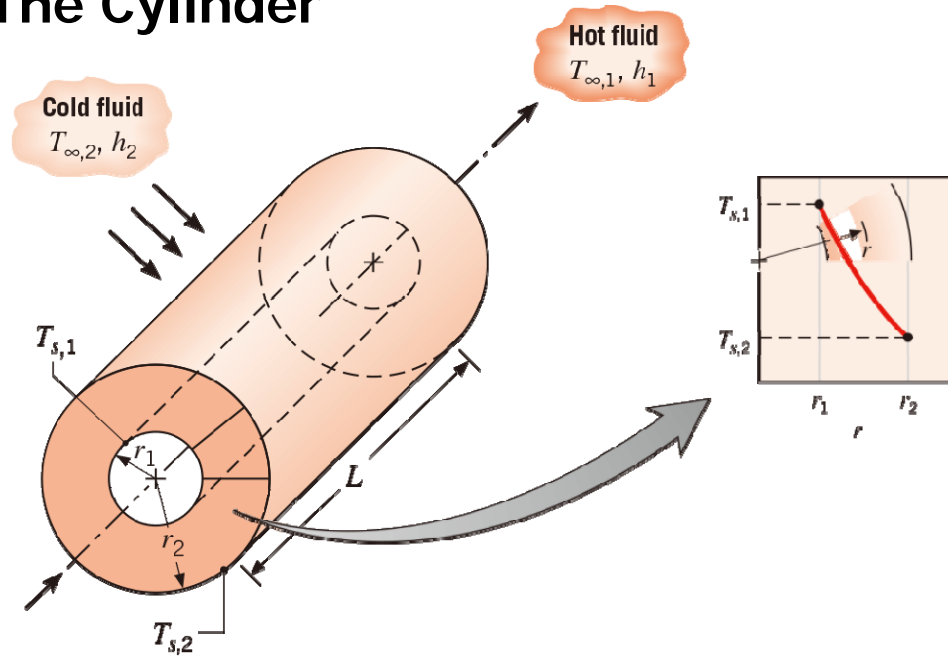
q_x may be computed by integrating between x_0 and x_1 .

A is uniform and k is independent of temperature, above equation reduces to

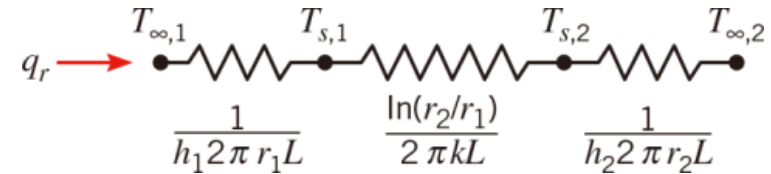
$$\frac{q_x \Delta x}{A} = -k \Delta T$$

3.3 Radial Systems

■ The Cylinder



$$q_r = -kA \frac{dT}{dr} = -k(2\pi rL) \frac{dT}{dr}$$



For steady-state conditions with no heat generation

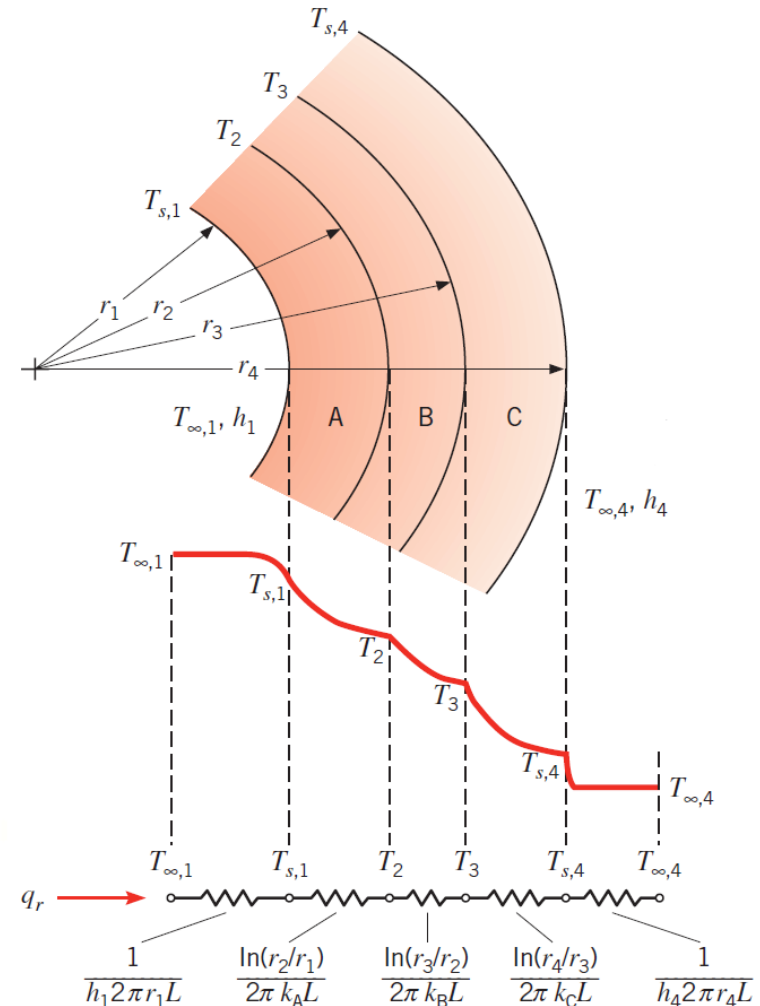
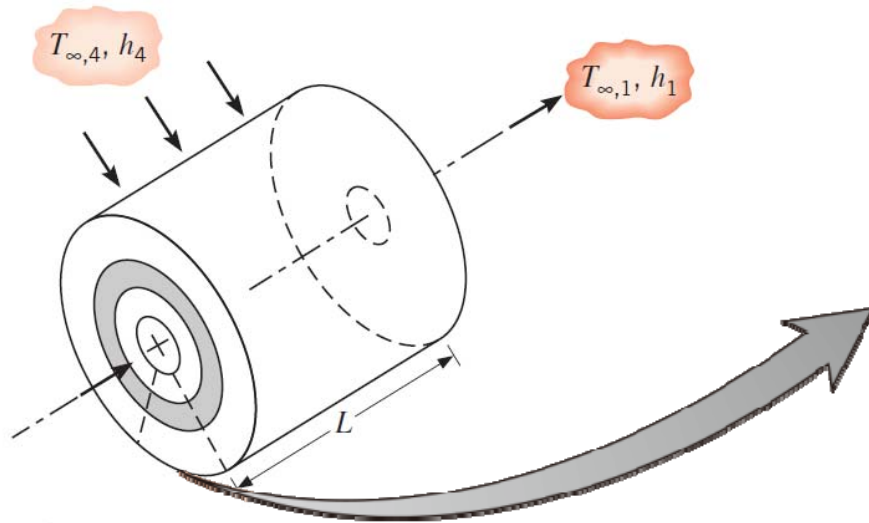
$$\frac{1}{r} \frac{d}{dr} \left(kr \frac{dT}{dr} \right) = 0$$

$$T(r_1) = T_{s,1} \quad \text{and} \quad T(r_2) = T_{s,2} \quad T_{s,1} = C_1 \ln r_1 + C_2 \quad T_{s,2} = C_1 \ln r_2 + C_2$$

$$T_{s,1} - T_{s,2} = C_1 \ln r_1 - C_1 \ln r_2 \quad C_1 = \frac{T_{s,1} - T_{s,2}}{\ln r_1 - \ln r_2} = \frac{T_{s,1} - T_{s,2}}{\ln \left(\frac{r_1}{r_2} \right)} \quad C_2 = T_{s,2} - \frac{(T_{s,1} - T_{s,2})}{\ln \left(\frac{r_1}{r_2} \right)} \ln r_2$$

3.3 Radial Systems

▪ Multilayered cylinder



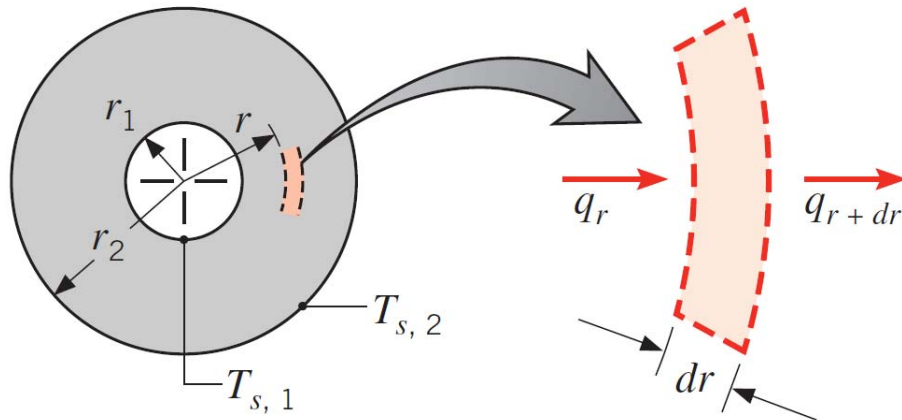
$$q_r = \frac{T_{\infty,1} - T_{\infty,4}}{\frac{1}{2\pi r_1 L h_1} + \frac{\ln(r_2/r_1)}{2\pi k_A L} + \frac{\ln(r_3/r_2)}{2\pi k_B L} + \frac{\ln(r_4/r_3)}{2\pi k_C L} + \frac{1}{2\pi r_4 L h_4}}$$

$$q_r = \frac{T_{\infty,1} - T_{\infty,4}}{R_{\text{tot}}} = UA(T_{\infty,1} - T_{\infty,4})$$

$$U = \frac{1}{\frac{1}{h_1} + \frac{r_1}{k_A} \ln \frac{r_2}{r_1} + \frac{r_1}{k_B} \ln \frac{r_3}{r_2} + \frac{r_1}{k_C} \ln \frac{r_4}{r_3} + \frac{r_1}{r_4} \frac{1}{h_4}}$$

3.3 Radial Systems

■ The Sphere



$$q_r = -kA \frac{dT}{dr} = -k(4\pi r^2) \frac{dT}{dr}$$

$$q_x \int_{x_0}^x \frac{dx}{A(x)} = - \int_{T_0}^T k(T) dT$$

$$q_r = \frac{4\pi k(T_{s,1} - T_{s,2})}{(1/r_1) - (1/r_2)}$$

$$R_{t,cond} = \frac{1}{4\pi k} \left(\frac{1}{r_1} - \frac{1}{r_2} \right)$$

■ Summary

	Plane Wall	Cylindrical Wall ^a	Spherical Wall ^a
Heat equation	$\frac{d^2T}{dx^2} = 0$	$\frac{1}{r} \frac{d}{dr} \left(r \frac{dT}{dr} \right) = 0$	$\frac{1}{r^2} \frac{d}{dr} \left(r^2 \frac{dT}{dr} \right) = 0$
Temperature distribution	$T_{s,1} - \Delta T \frac{x}{L}$	$T_{s,2} + \Delta T \frac{\ln(r/r_2)}{\ln(r_1/r_2)}$	$T_{s,1} - \Delta T \left[\frac{1 - (r_1/r)}{1 - (r_1/r_2)} \right]$
Heat flux (q'')	$k \frac{\Delta T}{L}$	$\frac{k \Delta T}{r \ln(r_2/r_1)}$	$\frac{k \Delta T}{r^2 [(1/r_1) - (1/r_2)]}$
Heat rate (q)	$kA \frac{\Delta T}{L}$	$\frac{2\pi Lk \Delta T}{\ln(r_2/r_1)}$	$\frac{4\pi k \Delta T}{(1/r_1) - (1/r_2)}$
Thermal resistance ($R_{t,cond}$)	$\frac{L}{kA}$	$\frac{\ln(r_2/r_1)}{2\pi Lk}$	$\frac{(1/r_1) - (1/r_2)}{4\pi k}$

3.5 Conduction with Thermal Energy Generation

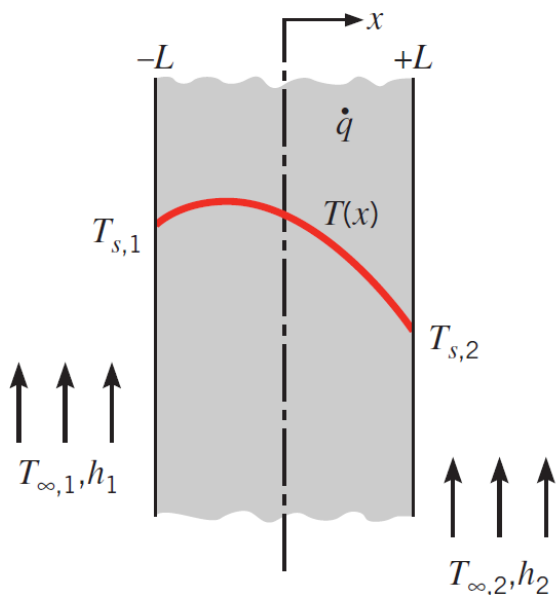
$$E_g = I^2 \cdot R_e$$

$$\dot{q} \equiv \frac{E_g}{V} = \frac{I^2 R_e}{V}$$

(W/m³)

- The **temperature distribution** of processes may be **occurring** within the medium due to thermal energy generation.
- A common thermal energy generation process involves the conversion from **electrical** to **thermal energy** in a **current-carrying medium** (Ohmic, or resistance, or Joule heating).
- Energy generation may also occur as a result of the **deceleration** and **absorption** of **neutrons** in the fuel element of a nuclear reactor or exothermic **chemical reactions**.

■ The Plane Wall



For constant thermal conductivity k and uniform energy generation per unit volume (\dot{q} is constant)

$$\frac{d^2 T}{dx^2} + \frac{\dot{q}}{k} = 0$$

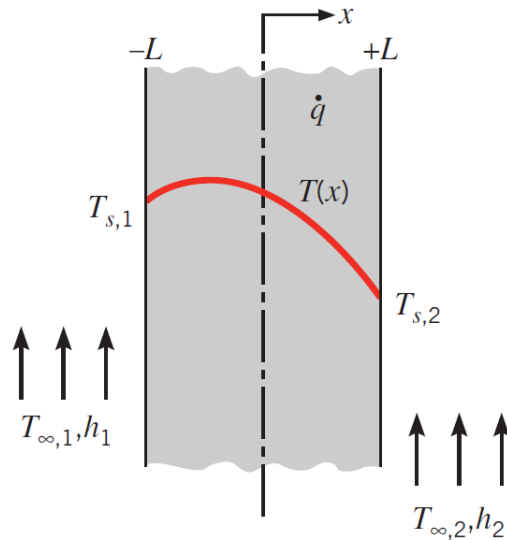
The general solution is

where C_1 and C_2 are the constants of integration.

C_1 and C_2 are determined through **boundary conditions**.

3.5 Conduction with Thermal Energy Generation

Asymmetrical Boundary Conditions



The general solution is $T = -\frac{\dot{q}}{2k}x^2 + C_1x + C_2$

$$T(-L) = T_{s,1} \quad T(L) = T_{s,2}$$

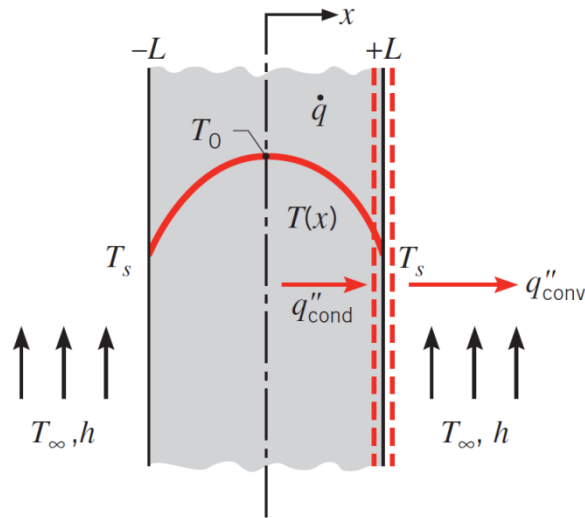
$$x = -L \quad T = T_{s,1} = -\frac{\dot{q}L^2}{2k} - C_1L + C_2$$

$$x = L \quad T = T_{s,2} = -\frac{\dot{q}L^2}{2k} + C_1L + C_2$$

$$T(x) = \frac{\dot{q}L^2}{2k} \left(1 - \frac{x^2}{L^2} \right) + \frac{T_{s,2} - T_{s,1}}{2} \frac{x}{L} + \frac{T_{s,1} + T_{s,2}}{2}$$

3.5 Conduction with Thermal Energy Generation

▪ Symmetrical Boundary Conditions



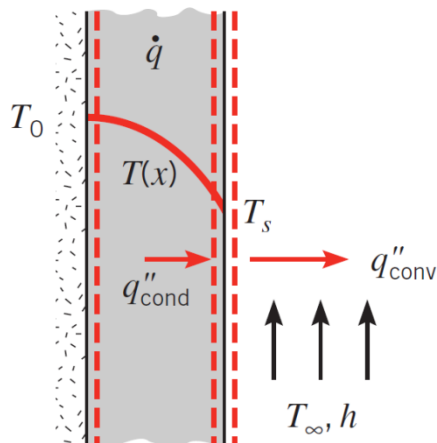
The general solution is $T = -\frac{\dot{q}}{2k}x^2 + C_1x + C_2$

$$T_{s,1} = T_{s,2} \equiv T_s$$

$$x = -L$$

$$x = L$$

$$T(x) = \frac{\dot{q}L^2}{2k} \left(1 - \frac{x^2}{L^2} \right) + T_s$$



The maximum temperature exists at the midplane

$$T(0) \equiv T_0 = \frac{\dot{q}L^2}{2k} + T_s$$